

2IB Physics HL only. Training May 2008 TM. Answers.

1. N02 H1: 12 - B

2. N02 S1: 14 - D

3. N02 H1: 18 - B

4. N02 S2: A1, H2: A1

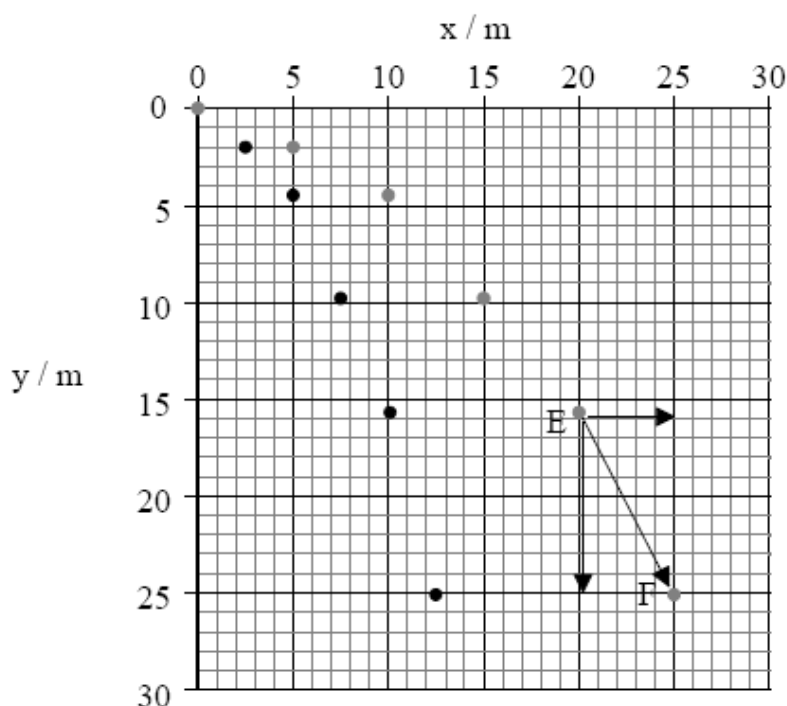
(a) horizontal: projectile moves 5 m in 0.5 s;

$$\text{so } v = \frac{5}{0.5} = 10 \text{ m s}^{-1};$$

(b) horizontal distance travelled between images is always the same;
so no significant atmosphere, since air resistance would otherwise slow the horizontal motion;

(c) different planet mass than Earth;
and different radius than Earth;

(d) displacement vector;
horizontal component vector;
vertical component vector;



(e) vertical: 9 m in 0.5 s;
 $v = 18 \text{ m s}^{-1};$

(f) half previous horizontal spacing;
same vertical positions at each time interval;

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5. M02 S2: B2

(ii) $\frac{p_1}{p_2} = \frac{T_1}{T_2}$ with $T_1 = 300 \text{ K}$;

to give $T_2 = 600 \text{ K}$;

Award [1] if K not used, 54 °C.

recognise that change in internal energy = $ms\Delta\theta$;

$$= 4.0 \times 10^{-3} \times 3.1 \times 10^3 \times 300;$$

$$= 3.7 \times 10^3 \text{ J};$$

335 J if 27 °C used, 7.4 × 10³ J if 600 K used – award [2], award [1] if 54 °C used – 670 J.

(c) (i) zero

(ii) $p_1V_1 = p_2V_2$;

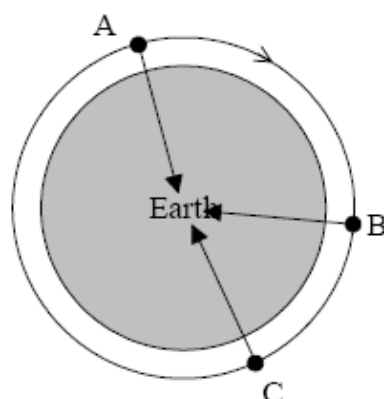
with $p_1 = 2.0 \times 10^5 \text{ Pa}$ and $V_1 = 2V_2$;

to give $p_2 = 4.0 \times 10^5 \text{ Pa}$;

(d) 3500 J

6. N02 H2: B4p1

(a)



vectors all towards the centre;
all the same length;

- (b) *Look for an argument along the following lines:*
a few hundred km is small compared to radius of the Earth;
 g depends on $\frac{1}{R^2}$;
so a small change in R will not produce a significant change in g ;
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- (c) $mg = \frac{mv^2}{R_E}$;
 $v^2 = gR_E$;
 $T = \frac{2\pi R_E}{v}$;
so $T = \frac{2\pi R_E}{\sqrt{gR_E}}$;
substitute $T = \frac{2 \times 3.14 \times 6.4 \times 10^6}{\sqrt{10 \times 6.4 \times 10^6}}$;
 $\approx 5000\text{s} \approx 84 \text{ min}$;

(d) $\frac{mv^2}{R} = \frac{GMm}{R^2};$

to give $v^2 = \frac{GM}{R};$

and $T = \frac{2\pi R}{v};$

therefore $T^2 = \frac{4\pi^2 R^2}{v^2} = \frac{4\pi^2 R^3}{GM};$

such that $\frac{R^3}{T^2} = \frac{GM}{4\pi^2} = \text{constant};$

(e) orbital period of geostationary satellite = $24 \times 60 = 1440$ min;

using $\frac{R^3}{T^2} = \text{constant}$ we have $\frac{R_E^3}{84^2} = \frac{(nR_E)^3}{1440^2};$

to give $n \approx 7;$

or $R_{\text{SAT}} \approx 7R_E;$